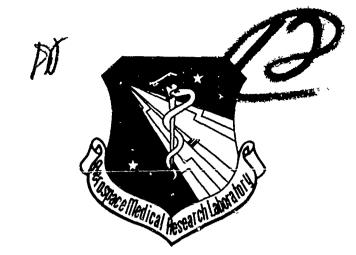
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USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

VOLUME 51 HH-53C IN-FLIGHT CREW NOISE

OCTOBER 1975

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AEROSPACE MEDICAL RESEARCH LABORATORY
AEROSPACE MEDICAL DIVISION
Air Force Systems Command
Wright-Patterson Air Force Base, Ohio 45433

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HENNING E. VON GIERKE

Director `

Biodynamics and Bionics Division

Aerospace Medical Research Laboratory

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	The HH-53C is a USAF heavy assault transport	helicopter used to search.
I	locate, and recover combat a recrew members, person	
1	hardware. This report provides measured data defi	ning the bioacoustic
ļ	environments at flight crew locations inside this	
l	flight operations. Data are reported for 15 locate physical and psychoacoustic measures: overall and	
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perceived noise level, and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Refer to Volume 1 of this handbook, USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application, AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

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PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 72310418, Measurement of Noise and Vibration Environments of Air Force Operations. Col Justus F. Rose, Jr. conducted the field measurements and performed the data analysis; Capt Nick Farinacci prepared this report.

The authors acknowledge the efforts of Mr. John N. Cole who established the data analysis requirements and assisted in the preparation of this report, and Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton who assisted in the mechanics of data processing.

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INTRODUCTION

The HH-53C is a USAF heavy assault transport helicopter used to search, locate, and recover combat aircrew members, personnel, and/or vital aerospace hardware. This helicopter, which is manufactured by the United Aircraft Corporation, Sikorsky Aircraft Division, is powered by two T64-GE-7 turboshaft engines rated at 3,925 shp at 13,600 rpm maximum power. The engines drive both a six-blade, fully-articulated, 22 m diameter main rotor, and a conventional four-blade 4.9 m diameter tail rotor. The engines are manufactured by the General Electric Company, Aircraft Engine Group, Military Engine Division.

This volume provides measured data defining the bioacoustic environments produced inside this helicopter. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with operations of the HH-53C helicopter.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and aerospace ground equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, aerospace ground equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. *Refer to Volume 1* (reference 1) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., in-flight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published, and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of the updated index as it is generated.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; Autovon 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

IN-FLIGHT NOISE

MEASUREMENTS

All noise measurements were made on-board a standard-configured HH-53C helicopter during typical speed, altitude, and flight maneuver conditions. This helicopter had full factory insulation as compared with those helicopters flown in SEA from which the insulation was removed. These levels describe the standard HH-53C environments, but may not be representative of those levels encountered if the helicopter has been configured differently (e.g., major equipment or structural changes).

Acoustic measurements were made at various flight crew locations. Table 1 lists the measurement locations and test conditions as numeric/alphabetic designators which are used on the data pages. The designator 1/A means measurement location 1 and test condition A.

The microphone position was at ear level external to headgear in a region 0.2-0.3 meters from the head when an individual was present. At unoccupied locations, measurements were made at ear level throughout a volume where the head would normally be located. In both cases the microphone was randomly moved throughout a spherical volume approximately 0.3 meters in diameter and the resultant samples analyzed using a 4- or 8-second integration time to obtain a power-averaged level that effectively smooths out short-duration fluctuations and best describes the exposure.

Although the presence of a crew member at a measurement location affects the resultant sound field, the magnitude of such effects is generally small and not significant in determining exposure limits or voice communication capabilities. Consequently, no distinction is made in this report between occupied and unoccupied measurement locations.

RESULTS

The measured data precented in Table 2 define the sound pressure levels (SPL) produced inside the HH-53C helicopter at the 15 specified locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data C-weighted and A-weighted sound levels, maximum permissible time for one exposure per day (AFR 161-35) with and without standard Air Force ear protectors, preferred speech interference level, and perceived noise level are calculated and presented in Table 3. These variety of measures are widely used to assess the effects of noise on personnel and their performance.

TABLE 1 MEASUREMENT LOCATIONS AND TEST CONDITIONS

HH-53C, Eglin AFB, 11 May 1971 Serial # 69-5789

LOCATION	POSITION	HEIGHT ABOVE DECK
1	Between Pilot and Copilot	Seated Head Level
2	Station 162 - Flight Engineers Station	Seated Head Level
3	Station 212, Centerline (1st row of Windows)	1.5 Meters
4	Station 212, Left Side	Seated Head Level
5	Station 222 Centerline	1.5 Meters
6	Station 302, Right Side	Seated Head Level
7	Station 312, Centerline (2nd row of Windows)	1.5 Meters
8	Station 312, Left Side	Seated Head Level

TABLE 1 (Continued)

MEASUREMENT LOCATIONS AND TEST CONDITIONS

HH-53C, Eghn AFB, 11 May 1971 Serial # 69-5789

LOCATION	POSIT	ion		HEIGHT ABOVE DECK
g	Station 342 (Directly pr	, Centerline ider gear box)		1.5 Meters
10	Station 412 of Windows	Centerline 12rd	row	1.5 Meters
11	Station 412,			
12				Seated Head Level
13	eage of ram		ard	1.5 Meters
	Station 500,			1.5 Meters
14	Just mside i	escue door, Righ	t Sido	
15	Gunners Sta	tion on ramp	or one	1.5 Meters 1.5 Meters
CONDITION	DI	ESCRIPTION		
A B	Internal AP	U running, ramp	down, crew	entrance door open.
D	Ground Idle	- ramp up, crew	entrance de	or alosed
	Tonous	" T PHRITIE		#2 Engine
	Torque Ng	10%		10%
	T5	65%		65%
	Rotor RPM	450°C		450°C
	Fuel Flow		50%	.00 0
	r det t 10W	250#/hr		250#/hr
C	Taxi Power -	- ramn un anom		
		- ramp up, crew a #1 Engine	entrance doo	r closed.
	Torque	#1 Engine 15%		#2 Engine
	Ng	80%		15%
	T5	475°C		80%
	Rotor RPM	410 C		475℃
	Fuel Flow	300#/hr	100%	-
n				300#/hr
D	Lift off/climb	- ramp up, crew	Antrones de	
		#1 Engine	cherance ap	or closed.
	Torque	80%		#2 Engine
	Ng	90%		80%
	T 5	500°C		90%
	Rotor RPM	_	100%	200°C
	Fuel Flow	1200#/hr	100%	1200#/hr
E	Cruica 1500	(D)		1200#/11]
	Or alse - 1000	PA, 138 KIAS -	- ramp up,	
	Torque	#1 Engine		#2 Engine
	Ng	55%		55%
	Tg	91%		91%
	Rotor RPM	560°C		560°C
	Fuel Flow	1000#/hr	100%	_
F				1000 <i>#/</i> hr
r	Same as E - r	amp down.		

3 OCTAVE											
NOISE SOURCE/SUBJECT:		OPERATI	ION		! !		; ; ;				0.1
HH-53C HELICOPTER Inflight Noise Levels) 17 JAN 75) PAGE F1
	1			ì		N/CON	1 14	1		1	!
FREQ (HZ)	1/8	1/6	1/E	2/A	2/8	3/E	7 H	5/A	6/B	9/9	
25	92	95	104	67	80	107	102	65	86	91	
S - T - T			2 0	9 9	77	9	90	63	92		
06			0	7.2	73	0	74	62	66		
63			97	80	77	0	0	7.	Q 9		
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160			96	89	94	66		92	91		
200			96	85	82	92		2.2	89		
250			76	8	85	32		78	න න		
315				D 9	7 to	# :		2 7 0 7	2 C		
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0 m			95	9.4	8 8	7.0		93	90		
8008			91	87	35	92		79	88		
1000			89	84	80	90		77	85		
1250			90	87	77	3 6		2.	91		
1600			89	19	77	91		73	60		
2000			98	9 0	8	88		~	83		
2500			98	49	77	8		69	89		
3150			3	7.7	9,	87		1	28		
0007			83	78	78	87		27	۲ (و		
5000			81	78	8	& 22		ù Z	بر م		
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16900 16003		5 2	82	4 4	7.8	8 4		92	9 6		
OVERALL	103	104	111	100	96	114	115	93	105	104	

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NOISE SOURCE/SUBJECT:		OPERATI	IONI		1		 	i 1 1 1	! ! !) TEST 71-010-001
HH-53C HELICOPTER						1					17 JAN 75
INFLIGH! NUISE LEVELS	·						1) PAGE F2
, 1	6/9	7/5	8/5	9/6	LOCATION	ON/CON	DITION	13/E	14/E	15/F	
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1250		96	96		φ 0 0 0 0				9 6		
1600		95	92	Š	91				97		
2000		88	83		88				97		
2500		91	93		91				96		
3150		83	6		88				36		
0007		86	87		82				96		
2000		82	8 2		83				36		
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1000) K	2 4		0 K				1 £		
12500		86	96		8 6				9.0		
16000		83	86		86				98		
OVERALL	108	113	114	111	114	117	115	115	115	117	
										1 1 1 1 1 1	

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE:	MEASURED SOUND OCTAVE BAND		PRESSURE LEVEL	(0B)		-					- - - - - - -) TOENTIFICATIONS
NOISE SOL HH-53C INFLIGH	NOISE SOURCE/SUBJECT: HH-53C HELICOPTER INFLIGHT NOISE LEVELS		OPERATION:	IONI		; ; ;) OHEGA 3.2 -) TEST 71-01G-001 RUH G1 17 JAN 75
FREQ (HZ)		1/8	1/6	1/E	2/A	L0CATI 2/8	LOCATION/CONDITION 2/8 3/E 4/E	DITION 4/E	5/A	6/8	9/9) PAGE J1
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OVERALL		103	104	111		1 2	P 0	ე დ	92	98	8 2	

ABLE: MEASURED SOUND 2 OCTAVE BAND	PRESSURE LEVEL	R LEVE	t (08)) IUENTIFICATION:
NOISE SOURCE/SUBJECT:		OPERATION:	10N1			^	1 1 2 2		!	1) OMEGA 3.2 -) TEST 71-016-001
HH-53C HELICOPTER INFLIGHT NOISE LEVELS											17 JAN 75
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31.5 63	102	106	107	106	107	107	105	107	108	44	
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